

A comparison of two methods of simulating the dynamic response of suspension seating

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# **1** Introduction

Theoretical models of suspension seats allow seat behaviour to be investigated without laboratory testing and without exposing human subjects to the high levels of vibration typical of suspension seat operating conditions. Various non-linear theoretical models of suspension seats have been described (e.g. Rakheja *et al.*, 1987, Gouw *et al.*, 1990, Rakheja *et al.*, 1994; Ranganathan and Sriram 1994; Wu and Griffin, 1995; Ahmed and Goupillon, 1997; Tewari and Prasad, 1999; Gunston, 2000; Rebelle, 2000). The majority of these models have used a lumped-parameter approach to describe the seat in terms of coefficients related to specific component parts and then solved the resulting non-linear equations using numerical integration techniques. The complexity of suspension seat models has progressively increased by including more detailed descriptions of the seat component parameters. The ISVR model, as described by Gunston (2000), continues this approach.

An INRS model, described by Rebelle (2000), adopted an alternative approach. The seat suspension and cushion were described using a Bouc-Wen model (Bouc, 1967; Baber and Wen, 1981) rather than in terms of specific coefficients for the individual components. The coefficients describing the seat dynamic behaviour were determined by curve-fitting to the measured behaviour of a specific seat. This approach allows a set of parameters to be

determined quickly for an existing seat from a small number of laboratory measurements, rather than by measuring each component individually.

This study tested two suspension seats in the laboratory and compared the measured seat performance with predictions made by the two alternative suspension seat models from ISVR and INRS. The models tested were the discrete parameter model developed at the Institute of Sound and Vibration Research (as described by Gunston, 2000) and the Bouc-Wen model developed at the Institut National de Recherche et de Sécurité (as described by Rebelle, 2000). The ISVR model was tested using measured damper coefficients and using an alternative set of coefficients optimised to the measured seat performance. The intention was to identify strengths and weaknesses of the alternative models.

## 2 The seats

Two production suspension seats were used for this investigation. The first was a compact design for use in industrial trucks with the suspension components mounted behind the backrest (Figure 1). The second seat was a model used in earthmoving machinery and had the suspension components mounted under the seat cushion (Figure 2). Both seats used covered foam cushions, steel coil springs, oil dampers and rubber end-stop buffers.





Figure 1 Schematic of the industrial truck seat

Figure 2 Schematic of the earthmover seat

## 3 The models

### 3.1 The INRS model

#### 3.1.1 Seat suspension model – Assumptions

To limit the number of parameters needed to model the behaviour of the seat, a global approach was considered, based on the measurement of the input motion (acceleration,

relative velocity or relative displacement) and output motion (acceleration, relative velocity or relative displacement) recorded, respectively, at the seat base and on the seat cushion.

Figure 3 shows a typical hysteretic behaviour of the seat measured with random excitation in a frequency range including the seat resonance frequency.



Figure 3: Force-displacement diagram - Seat exposed to random excitation.

The Bouc-Wen model describes a wide variety of hysteretic systems (Bouc, 1967). The friction force Z(t) is related to the relative displacement u(t) of the suspension according to the first order non-linear differential equation:

$$\dot{Z}(t) = (K - K_s)\dot{u} - \boldsymbol{g}\left|\dot{u}\right|Z - \boldsymbol{b}\dot{u}\left|Z\right|$$
(1)

where *K* and *K<sub>s</sub>* are positive stiffnesses and where *g* and *b* give the effect of the hysteresis. This model was first proposed by Bouc in 1967, then generalised and applied to structures under earthquake excitation by (Wen, 1976; Baber and Wen, 1981) and has been widely used for other applications since.

A non-linear single-degree-of-freedom vertical seat suspension with hysteresis can be modelled by two equations as follows:

$$\begin{cases} M\ddot{u} + C\dot{u} + K_{s}u + Z + F_{T} + F_{B} = -M\ddot{y} \\ \dot{Z} = (K - K_{s})\dot{u} - g|\dot{u}|Z - b\dot{u}|Z| \end{cases}$$
(2)

where u(t) is the vertical relative displacement of the suspension, Z(t) is the previously defined Bouc-Wen force, M is the mass, C is the viscous damping coefficient (assumed to be linear),  $K_s$  is the stiffness of the suspension, and  $\ddot{y}(t)$  is the input excitation.  $F_T$  and  $F_B$ represent the reaction force of the end-stop top buffers ( $F_T$  - Equation 3) and the bottom end-stop buffers ( $F_B$  - Equation 4). All the non-linear effects due to the different seat components are combined in the force, Z(t). Mass  $M (= m_i + m_{mp})$  represents the inert mass placed on the cushion ( $m_i$ ) and the mass of the moving parts of the seat suspension ( $m_{mp}$ ), both masses are assumed to be rigidly fixed to the seat suspension. Consequently, five parameters must be identified to describe the suspension behaviour (with no impact on the buffers) : C,  $K_s$ , K, g and b.

The top end-stop buffers were modelled simply as an equivalent buffer (i.e. a soft linear spring-damper,  $C_T$ ;  $K_T$ , system) which produces a reaction force,  $F_T$ , when the suspension exceeds its mid free-travel (*d*). The relative motion between the inert mass and the seat cushion were neglected. The inert mass was considered as fixed to the seat suspension model. On the other hand, this relative displacement was taken into account in the top buffer model to describe the acceleration time histories measured on the mass (see Section 3.1.2.2.2).

$$F_T = K_T (u-d) + C_T \dot{u} \qquad u > d \tag{3}$$

The bottom buffers were modelled as a cubic non-linear stiffness ( $K_B; K_{B1}$ ):

$$F_{B} = K_{B}(u+d) + K_{B1}(u+d)^{3} \qquad u < -d$$
(4)

In the model,  $F_T$  and  $F_B$  represent the reaction force of each end-stop buffer of the seat.

#### 3.1.2 Identification of the parameters

The parameters of the whole seat model were identified in two steps. Firstly, the suspension model parameters were obtained by means of curve fitting using the output acceleration time histories measured with no end-stop impacts. Secondly, the end-stop model parameters were also identified by curve fitting using the static force-deflection curves for the bottom buffer and using output acceleration measurements for the equivalent top buffer.

#### 3.1.2.1 Identification of the seat suspension model parameters

The test seats were mounted on a hydraulic shaker platform and excited by imposed random displacement signals over a frequency range from 0.5 Hz to 5 Hz. The excitation was generated during 16 seconds and included high and low levels of displacement amplitudes in order to generate all the seat behaviour phases: stick-slip phases and free motion. The absolute input acceleration was measured on the shaker platform. The relative output displacement and the absolute acceleration of the seat were also measured. Data acquisitions started with a zero level, for a few seconds, to fulfil null initial conditions in the

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numerical integration of the dynamic equations. The forklift seat was loaded with a 45 kg mass and the earthmover seat with a 58 kg mass  $(m_i)$ . The mass of the moving part of the seat was estimated at about 13.5 kg and 27 kg respectively  $(m_{mp})$ . The total moving mass,

 ${\it M}$  , was therefore equal to 58.5 kg and 85 kg.

The identification procedure was based on an optimisation method (SIMPLEX method, Nelder and Mead, 1965; or Levenberg-Marquardt method, Moré, 1977) minimising the quadratic error between the measurement and the numerical solution. The latter was obtained by the fourth-order Runge-Kutta method. The parameter values obtained are given in Table 1:

	$K_{s}$ (N/m)	$K~({\sf N/m})$	<b>g</b> (m <sup>-1</sup> )	<b>b</b> (m <sup>-1</sup> )	C (Ns/m)
Forklift	8156	320897	94371	-83727	301.3
Earthmover	5420	146620	87270	-84660	507

Table 1: Calculated parameter values of the seat suspension.

An example of the fit obtained from this calculation is shown in Figure 4:



Figure 4: Comparison between the calculated and measured output time histories. Seat loaded with a 45 kg mass  $(m_i)$ .

The main advantage of this approach is the simplicity of its application. In fact, to model any seat behaviour governed by the model defined in Equation 2 (not including impacts), the methodology required only two accelerometers: one for the input the other for the output. Moreover, this was achieved without dismantling the suspension components.

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#### 3.1.2.2 Identification of the end-stop model parameters

#### 3.1.2.2.1 Bottom buffer

The dynamic effects on the nominal bottom buffers were neglected. The end-stop buffers were modelled as a pure non-linear stiffness. The static force-deflection curve of the bottom buffers shown in Figure 5 was used to identify, by curve fitting, the model parameters given in equation (4):  $K_B$  and  $K_{B1}$ .



Figure 5: Comparison between modelled and measured buffer force-deflection curves. The values of  $K_{B}$  and  $K_{B1}$  are reported in Table 2.

#### 3.1.2.2.2 Top buffer

The aim was not to describe the real behaviour of the top buffers but to predict the amplitude of output acceleration peaks correctly when top impacts occur. Consequently, the relative displacement of the mass during the impact phases was not well described because the mass was considered as fixed to the seat. In practice, during high impacts, the mass is not always in contact with the seat cushion. The parameters of the impact force,  $F_T$ , were identified by fitting the whole model to the output acceleration response when impacts occurred. During this identification, all the seat parameters, including those of the bottom buffers, were kept at their previously identified values. Top buffer parameter values,  $C_T$  and  $K_T$ , were optimised from Equation 2. Table 2 gives the calculated values of the top and bottom buffer parameters.

	$K_B(N/m)$	$K_{B1} (N/m^3)$	$C_T (Ns/m)$	$K_T (N/m)$
Forklift	80000	3.4E8	200	9000
Earthmover	25020	1.46E8	0	6000

Table 2: Parameter values of top and bottom buffer model	(equation	(3) and	(4)).
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From the calculated result, the top buffers appeared to be soft, whereas their physical behaviour was experimentally found to be relatively stiff (in the static force-deflection curve). This result arises because the relative displacement between the inert mass and the seat cushion was neglected in the seat suspension system but was taken into account within an equivalent top buffer model.

For details concerning the validation of the model, referred to (Rebelle, 2000).

### 3.2 The ISVR model

#### 3.2.1 Model structure

The ISVR model is shown schematically in Figure 6. The equations describing each component can be found in Gunston (2000). The equations were solved by numerical integration using a 4<sup>th</sup> order Runga-Kutta algorithm.



#### Figure 6 Schematic of the ISVR mathematical model.

The cushion was described by a linear spring and damper element with a restriction that the overall cushion force acting downwards on the load could not cause the mass to exceed 1g downwards. This allowed the model to simply simulate the load lifting off the cushion. The suspension spring was described by a linear stiffness coefficient. The end-stop buffers were described by non-linear stiffnesses. The suspension damper force-velocity characteristic was included as a two-stage 3<sup>rd</sup> order polynomial curve fit to the measured damper force-velocity characteristic. Friction forces from the damper and the suspension linkage were modelled as constant forces opposing the motion or as a force just sufficient to prevent motion if the acting force was insufficient to overcome the friction force.

#### 3.2.2 Measurement of component parameters

#### 3.2.2.1 Overview

Describing a seat mathematically in terms of coefficient relating directly to the dynamic properties of the component parts required separate dynamic measurements of each of the components. However, this approach allowed the influence on the seat behaviour of a specific component to be investigated. The following sections describe briefly the measurements used to describe the seat components in terms of parameter values for use in the model. A summary of the parameter values used to describe the two seats can be found in Section 8 and a more detailed description of the measurements used to identify the seat components in terms of coefficient values can be found in Gunston (2000).

#### 3.2.2.2 The cushion

The cushions were preloaded to 500 N and subjected to a random motion band- limited at 1 Hz and 10 Hz. The applied acceleration and transmitted force were measured and converted to estimates of the cushion linear stiffness and damping using the following equations derived from Wei and Griffin (1998).

$$k(\mathbf{w}) = -\operatorname{Re}(M(\mathbf{w})) \cdot \mathbf{w}^{2}$$
(5)

$$c(\mathbf{w}) = -\operatorname{Im}(M(\mathbf{w})) \cdot \mathbf{w}$$
(6)

where k is the cushion stiffness, c is the cushion damping, and M is the cushion apparent mass. The estimated cushion stiffness, damping and coherency of the earthmover seat cushion are shown in Figure 7. The cushion stiffness varied by less than 10% over this range investigated. The cushion damping showed more variation over the frequency range. In both cases the values at 2 Hz were taken.

#### 3.2.2.3 The suspension linkage and stiffness

The suspension mechanisms were subjected to a quasi-static force-deflection test with the cushion and suspension damper mechanism removed. The suspension linkage force-deflection characteristic for the earthmover seat is shown in Figure 8. The suspension linkage friction was estimated as half the mean difference between the force in compression and extension over the free travel region between the end-stop buffers. The effective vertical suspension spring rate was estimated as the mean gradient over the same region.





Figure 7 Estimated cushion stiffness and damping

Figure 8 Suspension forcedeflection characteristic

#### 3.2.2.4 The suspension damper

The earthmover damper was tested on a commercial damper test rig that extracted the force-velocity characteristic and the friction in compression and extension from the measured damper response to sinusoidal excitations. The industrial truck damper was tested on an alternative apparatus. The force-velocity characteristic and friction values were again extracted from the response to sinusoidal excitations.

The friction value was estimated as the force transmitted by the damper at low velocities (<0.1 ms<sup>-1</sup>). The force-deflection characteristic was modelled as a two-stage 3<sup>rd</sup> order polynomial as shown in the following equations. Separate sets of coefficients were used to describe the friction and the force-velocity characteristic in compression and extension. The change in damping force with displacement due to the angled damper mounting used in the earthmover seat was accounted for.

$$F_{1} = c_{1}\dot{z} + c_{2}\dot{z}^{2} + c_{3}\dot{z}^{3} \bigg|_{\dot{z} < \dot{z}_{1}}$$
(7)

$$F_{2} = c_{4}(\dot{z} - \dot{z}_{1}) + c_{5}(\dot{z} - \dot{z}_{1})^{2} + c_{6}(\dot{z} - \dot{z}_{1})^{3} + c_{1}\dot{z}_{1} + c_{2}\dot{z}_{1}^{2} + c_{3}\dot{z}_{1}^{3} \bigg|_{\dot{z} \ge \dot{z}_{1}}$$
(8)  
$$F_{c} = F_{1} + F_{2}$$
(9)

where  $\dot{z}$  is the relative velocity across the damper along the damper axis,  $F_c$  is the damping force and  $\dot{z}_1$  is the velocity at which the damper changes between the polynomial described by coefficients  $c_1$ ,  $c_2$ ,  $c_3$  and that described by coefficients  $c_4$ ,  $c_5$ ,  $c_6$ .

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#### 3.2.2.5 The end-stop buffers

The end-stop buffers were described in terms of 5<sup>th</sup> order polynomial functions. The coefficients of these polynomials were determined by applying a least-squares curve fit to the measured buffer force-deflection characteristics.

### 3.2.3 Optimisation of damper parameters

As mentioned previously, the advantage of describing the seat in terms of individual parameters is that the effect of each component on the seat performance can be determined. The disadvantage of this approach is that a mathematical model and suitable parameter values must be obtained to describe every seat component part that affects the seat dynamic performance. Some of the seat components, such as the suspension damping and friction, were found to be difficult to measure.

Two sets of seat component parameters were used to describe each seat for the present investigation using the ISVR model. The first set was as measured by testing the individual components using the methods described in Gunston (2000). That study demonstrated that the damper coefficients could have a substantial influence on the predicted seat performance.

However, subsequent investigations (to be presented at the 2001 United Kingdom Conference on Human Response to Vibration) cast some doubt on the repeatability of measurements of the suspension damper coefficients, in particular the suspension friction. Repeated measurements of the same damper using different sets of apparatus and different methods resulted in substantially different friction values. The method of mounting the damper in the test rig and any resulting off-axis loads were suspected. Consequently, a simple downhill search algorithm was used to obtain a second set of coefficients for the suspension damper by curve fitting to measured data.

This process involved minimising the root-mean-square (r.m.s.) error between the measured and predicted seat load acceleration time histories using a gain applied to the force-velocity characteristic and a gain applied to the friction coefficients. The friction gain was optimised using a magnitude that did not cause end-stop impacts, and the force-velocity gain was obtained using a motion that caused severe end-stop impacts. The form of test motion as shown in Figure 9 was used in all cases, with a frequency of 2.35 Hz. The magnitudes of these motions are shown in Table 3. Different magnitudes were used for the different seats as the differing seat properties caused end-stop impacts to occur at different magnitudes.

The second set of coefficients was therefore identical to the first, apart from a gain applied to the force-velocity characteristic and a second gain applied to the friction coefficients. These gains are shown in Table 4.

Table 3 Seat base accelerations of the 2.35 Hz motions used to optimise the suspension damping parameters expressed in terms of r.m.s. acceleration and  $W_k$ -weighted VDV.

	Motion used to o friction coefficien impacts)	ptimise the ts (no end-stop	Motion used to optimise the damper force-velocity coefficients (end-stop impacts occurred)		
	r.m.s. accel	VDV	r.m.s. accel	VDV	
Earthmover seat	1.82 ms <sup>-2</sup>	1.62 ms <sup>-1.75</sup>	3.83 ms <sup>-2</sup>	3.42 ms <sup>-1.75</sup>	
Industrial truck seat	0.89 ms <sup>-2</sup> 0.78 ms <sup>-1.75</sup>		1.84 ms <sup>-2</sup>	1.63 ms <sup>-1.75</sup>	

Table 4 Optimal gains for the suspension damping coefficients as determined by the optimisation process

	Gain applied to coefficients	the	friction	Gain applied to the damper force-velocity characteristic coefficients.
Earthmover seat	0.61			0.70
Industrial truck seat	1.36	;		0.85

# 4 Experimental procedure

## 4.1 Laboratory seat tests

An earthmover seat and an industrial truck seat were tested in the laboratory at ISVR using a 58 kg rigid seat load. The seats were exposed to three frequencies of transient vibrations derived from vehicle cab floor motions measured in the field on off-road vehicles (Gunston, 1999). The acceleration waveform reproduced at the base of the seat is shown in Figure 9. The frequencies were 2.1 Hz, 2.35 Hz and 3.25 Hz and were chosen as being close to the peak of the power spectrum for the standard simulated vibration seat tests as shown in Table 5.



Figure 9 The input motion

Table 5 The approximate frequencies corresponding to the peak acceleration power spectra of standardised vibrations used for suspension seat testing

Approximate frequency of	Standard vibration	Intended vehicle	
the acceleration power	test		
spectrum peak value			
	ISO 7096 (2000) EM2	Scraper without suspension	
	ISO 7096 (2000) EM3	Wheel loader	
	ISO 7096 (2000) EM4	Grader	
	ISO 7096 (2000) EM5	Wheel dozer, wheeled soil	
2.1 Hz		compactor, backhoe loader	
	CEN/TC 231/WG9	All terrain industrial trucks	
	(1997) IT4		
	CEN/TC 231/WG9	Industrial trucks above 9000 kg	
	(1997) IT3		
	ISO/CD 5007 (1999)	Agricultural tractor of between	
2.35 Hz	Class 2	3600 and 6500 kg unballasted	
		mass	
3.25 Hz	CEN/TC 231/WG9	Industrial trucks between 3500 and	
5.25 112	(1997) IT2	9000 kg	

The seats were mounted on an electro-hydraulic shaker capable of  $\pm$ 500 mm peak displacement and  $\pm$ 10 ms<sup>-2</sup> peak acceleration. The seat index point device, as specified in ISO5353 (1999), was placed on the cushion and was ballasted to 58 kg using rigidly attached steel blocks. The load was placed on the seat at least 5 minutes before any measurements were taken. The driver mass control was adjusted so the mean ride position of the seat suspension was at the mid-point of the available free travel between the end-stop buffers. A 5 Hz  $\pm$ 1 mm sinusoid was used to overcome the suspension friction while adjusting the driver mass control.

The acceleration at the base of the seat and at the base of the load were measured using ENTRAN EGCSY-240D-10 accelerometers and the displacement of the seat suspension relative to the shaker platform was measured using an RDP DCT4000C LVDT. The signals from the transducers were acquired digitally at 512 samples per second using an *HVLab* data acquisition and analysis system via 100 Hz anti-aliasing filters. The acquired signals were low pass filtered at 40 Hz using a 6-pole zero-phase Butterworth filter to attenuate the predominantly 50 Hz measurement noise.

The input motions were generated over the range of magnitudes shown in Table 6 at 20 equally spaced intervals for each frequency. The performance of the seat in response to each motion was characterised using the ratio of the vibration dose value (VDV, see Equation 10) on the seat load to that recorded at the seat base. The VDV was calculated for each motion using the ISO2631  $W_k$  frequency weighting. The frequency weighting was applied using a digital filter including the effects of phase.

$$VDV = \left[\int_{t=0}^{t=T} a_w^4(t) dt\right]^{0.25}$$
(10)

where  $a_w$  is the W<sub>k</sub> frequency-weighted acceleration.

Frequency	Earthmover seat		Industrial truck seat		
	Lowest magnitude	Highest magnitude	Lowest magnitude	Highest magnitude	
2.1 Hz	0.81 ms <sup>-1.75</sup>	3.30 ms <sup>-1.75</sup>	0.43 ms <sup>-1.75</sup>	1.85 ms <sup>-1.75</sup>	
2.35 Hz	0.86 ms <sup>-1.75</sup>	4.20 ms <sup>-1.75</sup>	0.44 ms <sup>-1.75</sup>	1.96 ms <sup>-1.75</sup>	
3.25 Hz	1.05 ms <sup>-1.75</sup>	8.05 ms <sup>-1.75</sup>	0.47 ms <sup>-1.75</sup>	5.23 ms <sup>-1.75</sup>	

Table 6 The range of input magnitudes in terms of W<sub>k</sub> frequency-weighted VDVs.

## 4.2 Simulations

The acceleration time history recorded at the base of the seat was used as the input to the models. The predicted load mass acceleration was used to calculate a predicted VDV for comparison with the laboratory measurements.

Identical predicted and measured SEAT values would indicate that the amount of vibration, in terms of the VDV, was identical but would not necessarily indicate that the model was predicting exactly the same motion for the seat load. The normalised root-mean-square difference between the predicted and measured load mass acceleration, as defined by Equation 11, was also calculated to give an alternative indication of the accuracy of prediction of the acceleration waveform of the load mass.

$$e = \frac{\left[\int_{t=0}^{T} (a_{predicted}(t) - a_{measured}(t))^2 dt\right]^{\frac{1}{2}}}{\left[\int_{t=0}^{T} a_{measured}^2(t) dt\right]^{\frac{1}{2}}}$$
(11)

## 5 Results

The VDV on the seat load predicted from the models, compared with the laboratory measurements, is shown for all investigated frequencies of motion in Figure 10 to Figure 15. The corresponding r.m.s. errors between the predicted and measured seat load accelerations are shown in Figure 16to Figure 21.

## 6 Discussion

The following sections describe the conditions that resulted in differences in behaviour between the models.

### 6.1 The optimised versus non-optimised ISVR model

The ISVR model using measured values for the damper parameters can be seen to give the poorest results of the three models when compared with the measured results using the earthmover seat. The predictions of the VDV were consistently high in situations without end-stop impacts and consistently low with end-stop impacts, and the r.m.s. error was greater than the other two models with a difference of up to a factor of three at low magnitudes.

The differences between the predicted VDVs when using the industrial truck seat were smaller thanthose with the earthmover seat. The improvements in the r.m.s. error obtained by optimising the damper coefficients were also smaller with the industrial truck seat.

Further investigation of the methods of measuring seat components is required in order to more accurately specify specific components in terms of relevant coefficients. The comparatively good performance of the optimised model, in particular at low magnitudes, indicates that the existing theoretical model is capable of making good predictions of the seat dynamic performance when given a suitable set of coefficients.



Figure 10 Earthmover seat, 2.1 Hz input



Laboratory measurements ISVR predictions (non-optimised) ISVR predictions (optimised) INRS predictions E Load VDV (ms<sup>-1.75</sup>) R 8 8 8 .... 0.2 0.4 0.8 1 1.2 Seat base VDV (ms<sup>-1.75</sup>) 0.6 1.4 1.6 1.8

Figure 11 Industrial truck seat, 2.1 Hz input



1.75

Load VDV (ms'

Figure 12 Earthmover seat, 2.35 Hz input Figure 13 Industrial truck seat, 2.35 Hz input



Figure 14 Earthmover seat, 3.25 Hz input Figure 15 Industrial truck seat, 3.25 Hz input



Figure 16 Earthmover seat, 2.1 Hz input





Figure 17 Industrial truck seat, 2.1 Hz input



Figure 18 Earthmover seat, 2.35 Hz input







Figure 20 Earthmover seat, 3.25 Hz input Figure 21 Industrial truck seat, 3.25 Hz input

## 6.2 Low magnitude performance

The ISVR model resulted in the lowest errors at magnitudes that did not results in end-stop impacts for all frequencies with the earthmover seat, and for all frequencies except for 3.25 Hz with the industrial truck seat. The VDVs predicted by the ISVR and INRS models were similar over this range with the earthmover seat, and the INRS model gave better predictions with the industrial truck seat.

## 6.3 End-stop impact predictions for the earthmover seat

Both models underestimated the VDV on the earthmover seat in situations involving endstop impacts, and the r.m.s. error increased with increasing magnitude. Examination of the time histories showed that the models successfully predicted the acceleration due to the bottom end-stop impact, but did not predict a sufficiently severe upwards acceleration as the load returned to the seat after a top-stop impact.

## 6.4 Severe end-stop impacts with the industrial truck seat

The rate of increase of seat surface VDV with increasing magnitude measured in the laboratory with the industrial truck seat showed a tendency to decrease at high magnitudes. The ISVR model showed this trend while the INRS model did not.

## 6.5 The industrial truck seat at 3.25 Hz

The ISVR model overestimated the vibration with the industrial truck seat at 3.25 Hz. The INRS model gave better predictions and lower r.m.s. errors at all magnitudes, except for the lowest three magnitudes. The Bouc-Wen suspension model as used by the INRS model appears to be more suitable for modelling the seat suspension in this situation. Optimisation of the ISVR model damper parameters did not result in a large improvement in accuracy.

# 7 Conclusions

The two alternative models were found to give better performances in different circumstances. Improvements would be expected by refining the cushion models and the cushion-load interface to improve the simulation of top-stop impacts. Improved methods of characterising the seat suspension components (in particular the damper coefficients) would be beneficial to the ISVR model in order to obtain better predictions without optimising any seat component coefficients.

# 8 Parameters used in the ISVR model

			Earthmover	Industrial
	seat	truck seat		
Load mass	58 kg	58 kg		
Cushion stiffness	92.1 kNm⁻¹	170 kNm <sup>-1</sup>		
Cushion damping	1371 Nsm⁻¹	1250 Nsm <sup>-1</sup>		
Suspension moving mass			27 kg	13.5 kg
Suspension linkage friction			74 N	26 N
Suspension stiffness			4.57 kNm <sup>-1</sup>	10.4 kNm⁻¹
Suspension damper gas loading stiffn	ess		2.3 kNm <sup>-1</sup>	N/A
Horizontal distance between damper r mid ride	nounting points	at	150 mm	N/A
Vertical distance between damper mo ride	unting points at	mid	111 mm	N/A
		$c_1$	0	0
		<i>C</i> <sub>2</sub>	0	0
		<i>C</i> <sub>3</sub>	0	0
	Compression	$C_4$	-107	1.80x10 <sup>2</sup>
		$C_5$	6.32x10 <sup>3</sup>	0
		$c_6$	0	0
Damper force-velocity characteristic		$\dot{z}_1$	0	0
fit coefficients		$c_1$	0	-3.55x10 <sup>1</sup>
		$c_2$	0	1.50x10 <sup>3</sup>
		С3	0	0
	Extension	$C_4$	6.16x10 <sup>1</sup>	1.01x10 <sup>3</sup>
		$c_5$	1.03x10 <sup>4</sup>	3.49x10 <sup>3</sup>
		<i>c</i> <sub>6</sub>	0	1.44 x10⁵
		$\dot{z}_1$	4.5x10 <sup>-2</sup> ms <sup>-1</sup>	2.9x10 <sup>-1</sup> ms <sup>-1</sup>
Free travel between end-stops	64 mm	50 mm		
Damper friction in extension			173 N	30 N
Damper friction in compression			64 N	8 N
		$\cdot x^5$	1.83x10 <sup>11</sup>	3.20x10 <sup>12</sup>
Bottom buffor axial force deflection ch	oractorictic fit	$\cdot x^4$	-8.57x10 <sup>7</sup>	-8.04x10 <sup>10</sup>
coefficients where r is the buffer com		$\cdot x^3$	-4.45x10 <sup>7</sup>	8.72x10 <sup>8</sup>
	10331011	$\cdot x^2$	1.50x10 <sup>6</sup>	-3.86x10 <sup>6</sup>
		$\cdot x$	7.63x10 <sup>3</sup>	5.42x10 <sup>4</sup>
Number of bottom buffers			2	2
Horizontal distance between the ends of the linkage arm at			295 mm	N/A
Mid fide	150	N1/A		
Vertical distance between the ends of the linkage arm at mid ride			150 mm	N/A
Top buffer axial force-deflection characteristic fit coefficients where x is the buffer compression $ \frac{\cdot x^{5}}{\cdot x^{4}} $			5.48x10 <sup>15</sup>	0
			-2.57x10 <sup>4</sup>	8.2x10 <sup>11</sup>
			2.60x10 <sup>3</sup>	-6.66x10 <sup>9</sup>
			3.27x10 <sup>7</sup>	2.17x10 <sup>7</sup>
			1.35x10 <sup>5</sup>	3.98x10 <sup>4</sup>
Number of top buffers			2	2

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